



Analytical Modeling of Laser Moving Sources







Contains:

- Heat flow equation
- Analytic model in one dimensional heat flow
- Heat source modeling
 - Point heat source
 - Line heat source
 - Plane heat source
 - Surface heat source
- Finite difference formulation
- Finite elements



Heat flow equation

For developing basic heat flow equation, consider the differential element . Heat balance in element is given by,



Heat flow through differential element

Heat in – Heat out + Heat generated

= Heat accumulated

Heat in and out rates depends on conduction and convection.

For x axis,

Heat in- Heat (conduction)

$$-k\frac{\partial T}{\partial x} - \left(-k\left[\frac{\partial T}{\partial x} + \left(\frac{\partial (\partial T/\partial x)}{\partial x}\right)\Delta x\right]\right)\Delta y\Delta z$$

Heat in- Heat (convection or advection) $\rho C_p T U_x \Delta y \Delta z - \rho C_p U_x \left(T + \frac{\partial T}{\partial x} \Delta x\right) \Delta y \Delta z$



Heat Flow Equation

heat accumulation = $\rho C_p \frac{\partial T}{\partial t} \Delta x \Delta y \Delta z$

heat generated = $H\Delta x\Delta y\Delta z$

$$k\nabla^{2}T\Delta x\Delta y\Delta z - \rho C_{p}U\nabla T\Delta x\Delta y\Delta z + H\Delta x\Delta y\Delta z = \rho C_{p}\frac{\partial T}{\partial t}\Delta x\Delta y\Delta z$$

$$k\nabla^2 T - \rho C_p \frac{\partial T}{\partial t} - \rho C_p U \nabla T = -H$$



One dimensional heat conduction

If the heat flow in only one direction and there is no convection or heat generation, the basic equation becomes

$$\frac{\partial^2 T}{\partial z^2} = \frac{1}{\alpha} \frac{\partial T}{\partial t} \quad \text{where } \alpha = \text{diffusivity, t} = \text{time}$$

 Using separation of variables, the solution can be assumed to be a product of spatial variable, u(z), and time variable, v(t):

$$T(z,t) = u(z) v(t)$$

$$v(t) \frac{\partial^2 u(z)}{\partial z^2} = \frac{u(z)}{\alpha} \frac{\partial v(t)}{\partial t}$$

$$\frac{1}{u(z)} \frac{\partial^2 u(z)}{\partial z^2} = \frac{1}{\alpha v(t)} \frac{\partial v(t)}{\partial t} = -\beta^2$$

$$u(z) = A\cos(\beta z) + B\sin(\beta z)$$

$$v(t) = Ce^{-\alpha\beta^2 t}$$



Flux formulation in one dimension If the heat flow in only one direction with a flux input at z=0

$$\frac{\partial^2 T}{\partial z^2} = \frac{1}{\alpha} \frac{\partial T}{\partial t}$$
(1)
$$q(z,t) = -k \frac{\partial T(z,t)}{\partial z}$$
(2)

The formula is a second second

Differentiating Eq. (2) with respect to space variable

$$\frac{\partial^2 q}{\partial z^2} = -k \frac{\partial^3 T}{\partial z^3} \tag{3}$$

The one dimensional heat conduction equation given by Eq. (1) is differentiated with space variable,

$$\frac{\partial^3 T}{\partial z^3} = \frac{1}{\alpha} \frac{\partial}{\partial z} \left(\frac{\partial T}{\partial t} \right) \tag{4}$$

Differentiating Eq. (2) with respect to time variable yields,

$$\frac{\partial q}{\partial t} = -k \frac{\partial}{\partial t} \left(\frac{\partial T}{\partial z} \right)$$
(5)



Flux formulation

By manipulating Eq. (3) to (5)

$$\frac{\partial^2 q}{\partial z^2} = \frac{1}{\alpha} \left(\frac{\partial q}{\partial t} \right) \quad \text{in } 0 < z < \infty, t > 0 \tag{6}$$

The boundary and the initial conditions are given by,

$$q(z,t) = f_0 \text{ at } z = 0, t > 0$$

 $q(z,t) = 0 \text{ at } t = 0$

The solution for this is given by,

$$T(z,t) = \frac{2f_0}{k} \left[\left(\frac{\alpha t}{\pi}\right)^{\frac{1}{2}} e^{-\frac{z^2}{4\alpha t}} - \frac{z}{2} \operatorname{erfc}\left(\frac{z}{\sqrt{4\alpha t}}\right) \right]$$

At the surface (z=0) the temperature is

$$\mathsf{T(0,t)} = \frac{2f_0}{k} \left(\frac{\alpha t}{\pi}\right)^{\frac{1}{2}}$$



Heat source modelling: Introduction:

Why modeling?

1. Semi-quantitative understanding of the process mechanism for the design of experiments.

- 2. Parametric understanding for control purpose. E.g. statistical charts.
- 3. Detailed understanding to analyse the precise process mechanisms for the purpose of prediction, process improvement.

Types of heat sources:

Point heat source.

Line heat source.

Plane heat source. (e.g. circular , rectangular)



1.Instantaneous point heat source:

The differential equation for the conduction of heat in a stationary medium assuming no convection or radiation, is



gives the temperature rise at position (x, y, z) and time t due to an instantaneous heat source δq applied at position (x', y', z') and time t'; where δq = instantaneous heat generated, C = sp. heat capacity, α = diffusivity, ρ = Density, t = time, K = thermal conductivity.

Temperature rise in semi-infinite body

- A mirror image can be used
- No heat transfer at the surface of semi-infinite body





Semi-infinite body

- $dT(x, y, z, t) = \frac{\delta q}{\rho C (4\pi a(t-t'))^{3/2}} exp \left[-\frac{(x-x')^2 + (y-y')^2}{4a(t-t')} \right] \left[exp \left[-\frac{(z-z')^2}{4a(t-t')} \right] + exp \left[-\frac{(z+z')^2}{4a(t-t')} \right] \right]$
- If the heat is applied at the surface or z'=0, such as moving area heat sources

•
$$dT(x, y, z, t) = \frac{2\delta q}{\rho C (4\pi a (t-t'))^{3/2}} exp \left[-\frac{(x-x')^2 + (y-y')^2 + z^2}{4a (t-t')} \right]$$



Results:

- When (x, y, z) = (0,0,0) and (x', y', z') = (0,0,0)
 T = 473.3379
- 2. When (x, y, z) = (0.5,0.5,0) and (x', y', z') = (0,0,0) T = 413.0811
- 3. When (x, y, z) = (2,2,0) and (x', y', z') = (0,0,0) T = 53.5792

For temperature over entire surface consider heat source at (0,0,0) and workpiece have dimension 50 X 50. Temperature distribution Is shown in figure.



2. Continuous point heat source in infinite body:

If the heat is liberated at the rate dQ= P.dt' from t = t' to t = t'+ dt' at the point (x', y', z'), the temperature at (x, y, z) at time t is found by integrating above equation, and C = sp. heat capacity, α = diffusivity, ρ = Density. From the point heat source solution,

$$dT(x, y, z, t) = \frac{Pdt'}{\rho C(4\pi a(t-t'))^{\frac{3}{2}}} \exp\left[-\frac{(x-x')^2 + (y-y')^2 + (z-z')^2}{4a(t-t')}\right]$$

Now if the heat source is on from time t'=0 to t'=t continuously it can be written as

$$dT(x, y, z, t) = \int_{t'=0}^{t'=t} \frac{Pdt'}{\rho C (4\pi a (t-t'))^{\frac{3}{2}}} \exp\left[-\frac{(x-x')^2 + (y-y')^2 + (z-z')^2}{4a (t-t')}\right]$$



Continuous point heat source

To simplify the situation, one can assume that the heat source was switched on at time, t'=-t and turned off at t'=0

$$dT(x, y, z, t) = \int_0^t \frac{Pdt}{\rho C(4\pi a(t))^{\frac{3}{2}}} \exp\left[-\frac{(x - x')^2 + (y - y')^2 + (z - z')^2}{4a(t)}\right]$$

where Q is in Watts. As $t \rightarrow \infty$ steady state temperature distribution occurs given by

$$T(x, y, z) = \frac{P}{4\pi k} \exp[-(x - x')^2 + (y - y')^2 + (z - z')^2]$$



Moving heat source solution steps

- Moving heat source is in fact a continuous stationary source in moving frame of reference
- Next step is used to find the superposition of point solutions in spatial co-ordinates in moving frame of reference for obtaining, line, plane or volumetric heat source.
- Transform the solution to fixed coordinate system
- Integrate with respect to time (t') for final solution in T(x, y, z, t)





Moving point heat source in semi-infinite body

Moving laser source along X -axis in a semi -infinite body

In moving coordinate system:

$$dT(X,Y,Z,t) = \frac{2\delta q}{\rho C (4\pi a(t-t'))^{\frac{3}{2}}} \exp\left[-\frac{(X-x')^2 + (Y-y')^2 + (Z)^2}{4a(t-t')}\right]$$

In fixed coordinate system:

$$dT(x, y, z, t) = \frac{2\delta q}{\rho C (4\pi a(t-t'))^{\frac{3}{2}}} \exp[-\frac{(x-vt'-x')^2 + (y-y')^2 + (z)^2}{4a(t-t')}$$

Note that $\delta q = Pdt'$

Moving point heat source:

Consider point heat source P heat units per unit time moving with velocity v on semiinfinite body from time t'= 0 to t'= t. During a very short time heat released at the surface is dQ = Pdt'. This will result in infinitesimal rise in temperature at point (x, y, z) at time t given by,

$$dT(x, y, z, t) = \int_{t'=0}^{t'=t} \frac{2Pdt'}{\rho C(4\pi a(t-t'))^{\frac{3}{2}}} \exp\left[-\frac{(x-vt'-x')^2 + (y-y')^2 + (z-z')^2}{4a(t-t')}\right]$$

The total rise in of the temperature can be obtained by integrating from t'=0 to t'= t





Line heat source in infinite body:

Temperature for the line heat source can be obtained directly by integrating the solution of the point source in the moving coordinate system.

• line source in moving coordinate:

Line source parallel to z-axis and passing through point (x', y') in moving system. The temperature obtained by integrating , where C = sp. heat capacity, ρ = Density, K = thermal conductivity. Here QI = heat per unit length

For infinite body

$$dT'(X,Y,Z,t) = \frac{\delta q}{\rho C (4\pi a (t-t'))^{\frac{3}{2}}} \exp[-\frac{(X-x')^2 + (Y-y')^2 + (Z-z')^2}{4a (t-t')}]$$

This point source in moving coordinates can be superposed for infinite line along z,

$$dT'(X,Y,t) = \frac{q_l dt'}{\rho C (4\pi a(t-t'))^{\frac{3}{2}}} \int_{-\infty}^{\infty} \exp[-\frac{(X-x')^2 + (Y-y')^2 + (Z-z')^2}{4a(t-t')}] dz$$
 Fig. keyhole model (W. Steen)



fusion zone width w

traverse speed, v

keyhole

melt pool

Infinite line source

 Integrating in moving coordinate system with respect to spatial variables,

$$dT(X, Y, t) = \frac{q_l dt'}{4\pi k(t - t')} \exp\left[-\frac{(X - x')^2 + (Y - y')^2}{4a(t - t')}\right]$$

Convert to stationary frame and integrate to time
 X= x-vt', Y=y and Z=z

$$T - T_0(x, y, t) = \int_{t'=0}^t \frac{q_l dt'}{4\pi k(t - t')} \exp\left[-\frac{(x - vt' - x')^2 + (y - y')^2}{4a(t - t')}\right]$$

This can be integrated numerically



Moving line heat source

 Using the same concept used in stationary continuous point where the laser (heat source) started at t' =-τ, and at time τ=0 the laser source is at origin (x'=0 and y'=0). One can get solution at (X, Y) from laser source:

$$T(X, Y, t) = \int_0^t \frac{q_l \, d\tau}{4\pi k(\tau)} \exp\left[-\frac{(X + v\tau)^2 + Y^2}{4a(\tau)}\right]$$

Similar result can be obtained by transformation,

$$t - t' = \tau$$
$$x - \nu t' = x + \nu(\tau - t) = x - \nu t + \nu \tau$$

At time t, x-vt =X, location in moving or laser coordinate system

$$x - vt' = x - vt + v\tau = X + v\tau$$
 and $d\tau = -dt'$

The limits, at
$$t' = 0$$
, $\tau = t$ and $t' = t$, $\tau = 0$
 $T(X, Y, t) = \int_{t}^{0} -\frac{q_{l} d\tau}{4\pi k(\tau)} \exp\left[-\frac{(X + v\tau)^{2} + Y^{2}}{4a(\tau)}\right]$

This also needs to be integrated numerically



The steady state solution at $t \to \infty$,

$$T(x,y) = \frac{q_l}{2\pi k} e^{-\frac{v X}{2\alpha}} BesselK\left(0, \frac{v\sqrt{X^2 + Y^2}}{2\alpha}\right)$$

Bessel function of second kind 0 order

It may be noted that it is a steady-state solution and X, Yare from the laser center.



Plane heat source:

- Surface heat source:
- Area (circular, rectangular heat source)
- Applied on x-y plane.
- Temperature depends on intensity.
- Application: surface hardening, surface cladding etc.





Gaussian moving circular heat source:

Gaussian heat source intensity

$$I(x,y) = \frac{2*P}{\pi\sigma^2} \exp[-\frac{2(x^2+y^2)}{\sigma^2}]$$

In moving coordinate system,

$$dT(X,Y,Z,t') = \frac{2\delta q}{\rho C(4\pi a(t-t'))^{\frac{3}{2}}} \exp\left[-\frac{(X-x')^2 + (Y-y')^2 + (Z)^2}{4a(t-t')}\right]$$

$$dT = \frac{2dt'}{\rho C (4\pi a (t-t'))^{\frac{3}{2}}} I(x', y') dx' dy' \exp[-\frac{(X-x')^2 + (Y-y')^2 + (Z)^2}{4a (t-t')}]$$



Superposing the point solutions for the Gaussian beam,

$$dT(t') = \frac{4Pdt'}{\pi\sigma^2\rho C(4\pi a(t-t'))^{\frac{3}{2}}} \times \int_{-\infty}^{\infty} dx' \int_{-\infty}^{\infty} dy' \exp[-(\frac{2x'^2 + 2y'^2}{\sigma^2} + \frac{x'^2 - 2(X)x' + (X)^2 + y'^2 - 2Yy' + Y^2 + Z^2}{4a(t-t')})]$$

Moving heat source.

Where P = laser power, σ = beam radius, v = scanning velocity, a = diffusivity, t



Rewriting the solution for fixed coordinate system,

$$dT(t) = \frac{4Pdt'}{\pi\sigma^2\rho C(4\pi a(t-t'))^{\frac{3}{2}}} \frac{\pi\sigma^2 4a(t-t')}{\sigma^2 + 8a(t-t')} \exp\left[-\frac{2((x-vt')^2 + y^2)}{\sigma^2 + 8a(t-t')} - \frac{z^2}{4a(t-t')}\right]$$

$$T - T_0 = \frac{4P}{\rho C \pi \sqrt{4a\pi}} \int_{t'=0}^{t'=t} \frac{dt'(t-t')^{-0.5}}{\sigma^2 + 8a(t-t')} \exp\left[-\frac{2((x-vt')^2 + y^2)}{\sigma^2 + 8a(t-t')} - \frac{z^2}{4a(t-t')}\right]$$

Numerical integration can be carried out for the above equation

If solution is required in moving or laser coordinate system, transformation described in line source can be used: At time t, x-vt =X, location in moving or laser coordinate system $x - vt' = x - vt + v\tau = X + v\tau$ and $d\tau = -dt'$

$$T - T_0 = \frac{4P}{\rho C \pi \sqrt{4a\pi}} \int_0^t \frac{d\tau(\tau)^{-0.5}}{\sigma^2 + 8a(\tau)} \exp\left[-\frac{2(X + v\tau)^2 + Y^2}{\sigma^2 + 8a(\tau)} - \frac{Z^2}{4a(\tau)}\right]$$



Modeling Gaussian heat source:

Material and process parameters: for EN18 steel

Laser power = 1300W Scanning velocity = 100/6 mm/sec Interaction time = 0.18sec. Beam Radius = 1.5mm

Temperature distribution X-Y plane



Diffusivity = 5.1mm^2/sec Density = 0.000008 kg/mm^3 Sp. Heat capacity = 674 J/kg k

Temperature along X-Z plane.





Uniform intensity:

Uniform circular moving heat source:

In the Uniform heat source, Q is defined by the magnitude q and the distribution parameter σ . The heat distribution, Q, is given by, $I(x,y) = \frac{P}{A}$ Where $A = \pi^* \sigma^2$



for circular heat source integrating with space variables,

$$dT(X,Y,Z,t) = \frac{2Pdt'}{8\rho C \pi \sigma^2 (\pi a(t-t'))^{\frac{3}{2}}} \exp[-\frac{Z^2}{4a(t-t')}] \times \int_{-\sigma}^{\sigma} \exp[-\frac{(X-x')^2}{4a(t-t')}] dx' \int_{-\sqrt{\sigma^2-x'^2}}^{\sqrt{\sigma^2-x'^2}} \exp[-\frac{(Y-y')^2}{4a(t-t')}] dy'$$

Now final temperature equation is obtained by integrating with time from 0 to t,



•Uniform rectangular moving heat source:

Rectangular heat source of dimension -l < y' < l and -b < x' < b i.e. for moving with constant velocity v from time t' = 0 to t' = t. Heat intensity l is given by, $I(x,y) = \frac{P}{A}$ where A = 4*b*l

Integrating with the space variables,



$$dT(X,Y,Z,t) = \frac{2Pdt'}{4bl\rho C(4\pi a(t-t'))^{\frac{3}{2}}} \exp\left[-\frac{-Z^{2}}{4a(t-t')}\right]$$
$$\int_{-l}^{l} \exp\left[-\frac{(X-x')^{2}}{4a(t-t')}\right] dx' \int_{-b}^{b} \exp\left[-\frac{(Y-y')2}{4a(t-t')}\right] dy'$$

Results can be obtained by numerical integration with respect to time.



Comparison of Gaussian and uniform heat source: for EN 18 steel



Results:

Fig. Comparison of width/depth of hardened zone[13]

No	Distribution	Beam Shape	Max. Surface	Depth (mm)	Width (mm)
			Temperature(⁰ C)		
1	Gaussian	Circular	1272	1	1.5
2	Uniform	Circular	1072.3	0.68	1.5
3		Rectangular	996	0.55	1.5
		Long			
4		Rectangular	1015.2	0.12	1.65
		short			
5		Square	1023	0.5	1.2

Finite difference formulation:



- Nodal points
- Nodal network
- Regular or irregular
- Types coarser

- fine

convection and conduction









Finite Element Models



Thermal Modeling

- Heat generated in workpiece due to cutting is small compared to the heat generated by the laser
- A scaled model (5mm x 2mm x 2mm) is used
- The Gaussian distribution of laser power intensity $P_{x,y}$ is given by: $P_{x,y} = \frac{2P_{tot}}{\pi r_b^2} exp\left(-\frac{2r^2}{r_b^2}\right)$
- The average absorptivity of incident irradiation is determined experimentally
- Temperature dependent thermophysical properties are used



Mathematical Formulation

The 3-D transient heat conduction equation is given by,

$$\frac{\partial}{\partial x}\left(k\frac{\partial T}{\partial x}\right) + \frac{\partial}{\partial y}\left(k\frac{\partial T}{\partial y}\right) + \frac{\partial}{\partial z}\left(k\frac{\partial T}{\partial z}\right) + \dot{Q} = \rho c_p \frac{\partial T}{\partial t} + \rho c_p V \frac{\partial T}{\partial x}$$

Initial condition, $T(x, y, z, 0) = T_0$

Natural boundary condition on front face,

$$k \frac{\partial T}{\partial n} - q + h(T - T_0) + \sigma \varepsilon (T^4 - T_0^4) = 0$$
$$k \frac{\partial T}{\partial n} - q + h_e(T - T_0) = 0$$
$$h_e = 2.4 \times 10^{-3} \varepsilon T^{1.61}$$



Mathematical Formulation

- Average measured temperatures are used for boundary conditions on remaining external surfaces
- Half symmetry used at bottom face

 $q_{bottom} = 0$



Case Study- Thermal Model



- Mapped dense mesh (25 μ m x 12.5 μ m x 20 μ m)
- An 8 noded 3-D thermal element (Solid70) is used
- Gaussian distribution of heat flux applied to a 5x5 element matrix which sweeps the mesh on the front face

Temperature Simulation



